Detecting weak physical signals from noise: A machine-learning approach with applications to magnetic navigation

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Magnetic Navigation - An Alternative to GPS Navigation







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Original Article 🛛 🔂 Full Access

Absolute Positioning Using the Earth's Magnetic Anomaly Field[†]

 ${\small \bullet} \hspace{0.1 cm} \textbf{Correction(s) for this article} \hspace{0.1 cm} \hspace{0.$

Aaron Canciani, John Raquet

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- Robust aerial navigation without GPS is challenging;
- Alternative navigation systems for airborne use are often limited by where and when they can operate (e.g., terrain following systems cannot operate over oceans; startracker and computer-vision systems depend on weather and daylight)
- The Earth's magnetic anomaly field is globally available at all times, which has the potential to overcome many of the limitations and can be exploited for non-GPS based aerial navigation.

Airborne Magnetic Anomaly Navigation

January 2017 · IEEE Transactions on Aerospace and Electronic Systems DOI: 10.1109/TAES.2017.2649238

Aaron Canciani · John Raquet



Generated from inside the earth:

- Core field 20-60 mT: dominant, responsible for the working of compasses
- Anomaly field about 100 nT: due to the permanent or induced magnetization of the rocks in the earth's crust

Key feature of the anomaly field:

- The strength of the anomaly field depends on the location possible for positioning and navigation
- When collected from, e.g., an airplane, the anomaly field is effectively a time series signal

Generated outside of the earth:

• Temporally varying field - about 10 nT: from the ionosphere, magnetosphere, and the coupling currents between the two



NASA Earth Observatory, "Measuring Earth's Magnetism," 2014. https://earthobservatory.nasa.gov/images/84266/ measuring-earths-magnetism



U.S. Geological Survey, "Magnetic anomaly maps and data for North America," 2021. https://mrdata.usgs.gov/magnetic/



Total measured field: $\vec{B}_m = \vec{B}_e + \vec{B}_{aircraft} = \vec{B}_{core} + \vec{B}_{anomaly} + \vec{B}_{tv} + \vec{B}_{aircraft}$

- \vec{B}_{core} calculated from the International Geomagnetic Reference Field (IGRF) coefficients
- \vec{B}_{tv} mostly from the diurnal variations and space weather can be removed using ground-based reference measurements
- $\vec{B}_{aircraft}$ total field generated by the aircraft



- Calibration: Tolles-Lawson (TL) model to estimate $\vec{B}_{aircraft}$
- Applied to the reading of the magnetometer at the tail stinger \rightarrow real value of the earth field \vec{B}_e
- Subtracting \vec{B}_{core} and \vec{B}_{tv} from \vec{B}_e gives the anomaly field $\vec{B}_{anomaly}$

 $|\vec{B}_{\rm e, TL}| = |\vec{B}_{\rm m}| - B_{\rm TL}|$



TL model: the magnetic field generated by the body of the aircraft as three magnetic moments:

- 1. Permanent magnetic moment: the nearly constant magnetic moment of the entire aircraft.
- 2. Induced magnetic moment: the magnetic response of the magnetically susceptible materials in the aircraft to the earth magnetic field.
- 3. Eddy current moment: caused by the temporal variations of the earth magnetic field due to the motion of the aircraft.

$$B_{\rm TL} = B_{\rm perm} + B_{\rm ind} + B_{\rm eddy}$$

= $\vec{B}_{\rm m}^T \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + |\vec{B}_{\rm m}| \vec{B}_{\rm m}^T \begin{bmatrix} x_4 & x_5 & x_6 \\ x_5 & x_7 & x_8 \\ x_6 & x_8 & x_9 \end{bmatrix} \vec{B}_{\rm m}$
+ $|\vec{B}_{\rm m}| \vec{B}_{\rm m}^T \begin{bmatrix} x_{10} & x_{11} & x_{12} \\ x_{13} & x_{14} & x_{15} \\ x_{16} & x_{17} & x_{18} \end{bmatrix} \dot{\vec{B}}_{\rm m},$

- x_1, \dots, x_{18} : 18 constant coefficients that can be calculated after a calibration flight
- TL model works well only when the flying aircraft is in a magnetically quiet mode and all the magnetometer measurements are performed on a tail stinger outside the cabin
- Often this is not the case!



• Recall:

$$\vec{B}_e = \vec{B}_{core} + \vec{B}_{anomaly} + \vec{B}_{tv}$$

• TL calibration gives an estimate of \vec{B}_e :

$$\hat{\vec{B}}_e = \vec{B}_{e,TL}$$

- \vec{B}_{core} calculated with the International Geomagnetic Reference Field (IGRF) coefficients
- \vec{B}_{tv} mostly from the diurnal variations and space weather can be removed using ground-based reference measurement

$$\rightarrow \vec{B}_{anomaly} \approx \vec{B}_{e,TL} - \vec{B}_{core} - \vec{B}_{tv}$$

- *TL calibration works well only when the flying aircraft is in a magnetically quiet mode and all the magnetometer measurements are performed on a tail stinger outside the cabin.*
- For normal flights, these conditions are not met.
- Magnetic signals collected inside the cockpit are noisy due to the electronics a weak signal embedded in overwhelmingly strong noise!
- Use TL model to obtain the ground truth for training neural networks

Data Source: USAF-MIT Artificial Intelligence Accelerator



i≣ README.md

Signal Enhancement for Magnetic Navigation Challenge Problem

Sgitlab

This is a repository for the signal enhancement for magnetic navigation (MagNav) challenge problem, which was introduced at JuliaCon 2020. The high-level goal is to use magnetometer (magnetic field) readings recorded from within a cockpit and remove the aircraft magnetic noise to yield a clean magnetic signal. A detailed description of the challenge problem can be found here and additional MagNav literature can be found here.

Round	Start	End	Winning Team
1	26-Jul-20	28-Aug-20	Ling-Wei Kong, Cheng-Zhen Wang, and Ying-Cheng Lai Arizona State University (submission)
2	24-Sep-20	31-Jan-21	

Introduction Videos

- Magnetic Navigation Overview
- Challenge Problem Description
- Challenge Problem Datasets

Starter Code

A basic set of starter Julia code files have been provided within the src folder. This code is largely based on work done by Major Canciani. This code has only been tested with Julia 1.4 and 1.5. A sample run file is located within the runs folder, which includes downloading the flight data via artifact (Artifacts.toml). Details of the flight data are described in the readme files within the readmes folder. The flight data can also be directly downloaded from here.

Team Members

The MagNav team is part of the USAF-MIT Artificial Intelligence Accelerator, a joint collaboration between the United States Air Force, MIT CSAIL, and MIT Lincoln Laboratory. Current team members include:



Data Structure



- Source: Open Call for developing machine-learning approaches to signal enhancement for magnetic navigation (MagNav) Challenge organized in 2020 by the Air Force-MIT Artificial Intelligence Accelerator
- Four other magnetometers placed at different positions in the cabin
- 45 available flight lines







github.com/MIT-AI-Accelerator/MagNav.jl

Examples



Line number	Description	Training length (s)	Validation/Test length (s)
1003.02	Eastern Ontario Free-Fly 400m	2246.5	748.8
1003.03	Climb to 800m	61.3	20.4
1003.04	Eastern Ontario Free-Fly 800m	2877.7	959.2
1003.05	Transit at 800m	246.7	82.2
1003.06	Descend to 400m	83.5	27.8
1003.07	Transit to Renfrew Free-Fly	58.9	19.6
1003.08	Renfrew Free-Fly 400m	2581.9	860.6
1003.09	Climb to 800m	82.3	27.4

TABLE II. Line number summary for flight 1003

TABLE V. Line number summary for flight 1006

Line number	Description	Training length (s)	Validation/Test length (s)
1006.03	Climb to 17,000ft	448.3	149.4
1006.04	Compensation maneuvers at 17,000ft	2547.7	849.2
1006.05	Descent to 10,000ft	317.5	105.8
1006.06	Compensation maneuvers at 10,000ft	369.1	123.0
1006.07	Transit/Descent to Eastern Ontario	732.1	244.0
1006.08	Compensation maneuvers in Eastern Ontario at 400m	479.5	159.8



61 other features recorded at the same time by some current and voltage sensors and readings from the INS system

Examples of some "most important" features:

•

Z.-M. Zhai, M. Moradi, L.-W.
Kong, and Y.-C. Lai, "Detecting weak physical signal from noise:
A machine-learning approach with applications to magnetic-anomaly guided navigation," *Physical Review Applied* 19, 034030, 1-18 (2023).

TABLE I. Importance ranking of the features selected by a greedy algorithm					
Features	Units	Description			
flux_c_t	nT	Flux C: fluxgate total			
cur_ac_lo	Α	Current sensor: air conditioner fan low			
${ m ins_alt}$	m	INS computed elevation			
flux_c_z	nT	Flux C: fluxgate z axis			
$flux_a_t$	nT	Flux A: fluxgate total			
vol_back_p V Voltag		Voltage sensor: resolver $board(+)$			
vol_back_n	V	Voltage sensor: resolver board(-)			
ins_lat	rad	INS computed latitude			
cur_com_1	Α	Current sensor: aircraft radio 1			
flux_c_y	nT	Flux C: fluxgate y axis			
vol_acpwr	V	Voltage sensor: aircraft power			
ins_wander	rad	INS computed wander angle			
cur_flap	Α	Current sensor: flap motor			
vol_bat_2	Α	Current sensor: battery 2			
ins_roll	\deg	INS computed aircraft roll			

Fluxgates B, C, D along x, y, and z axes have been used in TL calibration (Air Force – MIT MagNav Challenge)





github.com/MIT-AI-Accelerator/MagNav.jl



Determining the number of features as additional input signals (greedy algorithm)



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Machine Learning Scheme 1: Reservoir Computing





L.-W. Kong, Y. Weng, B. Glaz, M. Haile, and Y.-C. Lai, "Reservoir computing as digital twins of nonlinear dynamical systems," *Chaos* **33**, 033111, 1-21 (2023)

Reservoir Computing for Weak Signal Detection





- Recurrent neural network architecture
- Input signal is always available: a filtering problem
- A large (usually complex) network in a single hidden layer

Z.-M. Zhai, M. Moradi, L.-W. Kong, and Y.-C. Lai, "Detecting weak physical signal from noise: A machine-learning approach with applications to magnetic-anomaly guided navigation," *Physical Review Applied* **19**, 034030, 1-18 (2023).

Reservoir Computing: Representative Results









• A few hidden layers, each with a small neural network

- Memory property realized by delaycoordinate embedding
- High computational efficiency

ARTICLE

https://doi.org/10.1038/s41467-021-25801-2 OPEN

Next generation reservoir computing

Daniel J. Gauthier^{® 1,2™}, Erik Bollt^{3,4}, Aaron Griffith^{® 1} & Wendson A. S. Barbosa^{® 1} NATURE COMMUNICATIONS | (2021)12:5564 | https://doi.org/10.1038/s41467-021-25801-2



Determining the embedding dimension







An Alternative Machine-Learning Approach



A Decision Tree	Example of features:	TABLE I. Importance ranking of the features selected by a greedy algorithm		
		Features	Units	Description
		flux_c_t	nT	Flux C: fluxgate total
$F_1 > F_1$	$F_1 > \tilde{F}_1$			Current sensor: air conditioner fan low
YN			m	INS computed elevation
			nT	Flux C: fluxgate z axis
			nT	Flux A: fluxgate total
		vol_back_p		Voltage sensor: resolver $board(+)$
$F_2 > \tilde{F}_2$	$F_2 > \tilde{F}_2$	vol_back_n	V	Voltage sensor: resolver board(-)
	$\Gamma_2 > \Gamma_2$	ins_lat	rad	INS computed latitude
	N	cur_com_1	A	Current sensor: aircraft radio 1
Y N	Y N	flux_c_y	nT	Flux C: fluxgate y axis
		vol_acpwr	V	Voltage sensor: aircraft power
	~ (n , ñ)	ins_wander		INS computed wander angle
$F_3 > \tilde{F}_3$ $F_3 > \tilde{F}_3$ F_3	$> \tilde{F}_3$ $F_3 > \tilde{F}_3$	cur_flap	A	Current sensor: flap motor
		vol_bat_2	A	Current sensor: battery 2
		ins_roll	deg	INS computed aircraft roll
$\bullet \bullet \bullet \bullet \bullet \bullet \bullet \bullet$	$\bullet \bullet \bullet \bullet \bullet$			
			Par	ameters to be trained:
			1 41	
				$\tilde{F}_1, \tilde{F}_2, \tilde{F}_3, \dots$

Labels: a large collection of possible values of the weak signal (to be detected) in a suitable range (values in the training set + random values)

Random Forest





Random Forest:

- An ensemble of decision trees;
- Each tree is trained using a different subset of data

Training:

- Training a tree with a randomly selected set of features and a fraction of the available training data;
- Adding the trained tree to the "Forest"
- Adding more trained trees to the "Forest"

Prediction:

• Combining all predictions from the trees in the "Forest"

A. Moradi, Z.-M. Zhai, A. Nielsen, and Y.-C. Lai, "Random forest for accurate detection of weak physical signals," working paper (2023)

Parameter Tracking



A chaotic food-chain





Inverse Problem











During testing or deployment:

- Parameters are no longer accessible
- Their variations are unknown

Machine-learning input

• Partial state observations are available – machine-learning inputs



Machine-learning scheme: Adaptable Reservoir Computing – Why?



- The relation between the error in the detected earth's anomaly magnetic field and positioning precision is nonlinear.
- An error below 6.5 nT corresponds to the positioning error of less than 45 m.
- Empirically, the position error is approximately about 10 40 m when the magnetic signa error is around 4 nT.
- The mean magnetic signal errors from both reservoir computing and time-delayed feed forward neural networks is about 4 nT
- The anomaly field detected by reservoir computing and feedforward neural networks can be used for actual aircraft navigation positioning.
- Alternative machine-learning method: Random Forest

Ongoing work:

Developing **Transfer Learning** methods to deal with the situations where tail stinger measurements are not available - collaboration with Dr. Aaron Nielsen from Air Force Institute of Technology



- Least squares fitting e.g., E. W. Weisstein, Least squares fitting, https://mathworld.wolfram.com/ (2002)
- Maximum likelihood estimation e.g., J.-X. Pan and K.-T. Fang, Maximum likelihood estimation, pp. 77-158 in Growth Curve Models and Statistical Diagnostics (Springer, 2002)
- Bayesian estimation e.g., A. J. Haug, Bayesian Estimation and Tracking: A Practical Guide (John Wiley & Sons, 2012)
- Genetic algorithm e.g., L. Yao and W. A. Sethares, Nonlinear parameter estimation via the genetic algorithm, IEEE Trans. Signal Proc. 42, 927 (1994)
- Neural networks e.g., P. Guo, M. R. Lyu, and C. L. P. Chen, Regularization parameter estimation for feedforward neural networks, IEEE Trans. Sys. Man Cyber. B 33, 35 (2003)
- Markov chain Monte Carlo e.g., F. Yandun, M. Torres-Torriti, and F. Auat Cheein, Markov chain Monte Carlo parameter estimation for nonzero slip models of wheeled mobile robots: A skid steer case study, J. Mech. Robot. 13 (2021)
- Kalman filter e.g., G. Evensen, The ensemble Kalman filter for combined state and parameter estimation, IEEE Cont. Sys. Mag. 29, 83 (2009); L. Ljung, Asymptotic behavior of the extended Kalman filter as a parameter estimator for linear systems, IEEE Trans. Auto. Cont. 24, 36 (1979).



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Limitation: full-state measurements – time series of all dynamical variables of the system are required